

# AVIATION

*The Oldest American Aeronautical Magazine*

JULY 28, 1924

Issued Weekly

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The Navy SDW long range scout plane (Wright T3 engine) which has a cruising range of 2400 mi.

VOLUME  
XVII

## SPECIAL FEATURES

NUMBER

4

PLANS FOR SCHNEIDER CUP RACE  
PROGRESS OF AMERICAN WORLD FLIGHT  
NEW POLICY OF THE JOINT AERONAUTICAL BOARD  
FITTINGS AND OTHER STRUCTURAL PARTS OF AIRPLANES

GARDNER PUBLISHING CO., Inc.  
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JULY 28, 1924

# AVIATION

VOL. XVII, NO. 4

Published every Monday

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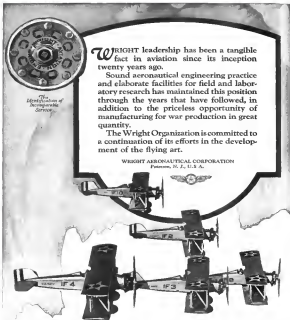
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## AVIATION

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Vol. XXIV

JULY 28, 1924

No. 4

### Championing a National Air Policy

**O**NE of the principal difficulties of getting a definite National Air Policy has been the lack of anyone in governmental circles who felt that it was a part of his duty. In Great Britain, France and Italy there are men of ability in the government who are charged with the responsibility for a national air policy.

In the United States, the Secretary of War is only concerned with aircraft as a comparatively small part of Army activity. The Secretary of the Navy regards the Bureau of Aeronautics in the light of an "adjutant" to the fleet activities. The Secretary of Commerce, whose the Wagner Bill would place in charge of civil aviation, probably regards his possible supervision of commercial aviation as he would under law. The Post Master General has in the Air Mail a most spectacular triumph of postal efficiency and permission. The Secretary of State feels that international aviation involve new international problems that are difficult to solve independently of the League of Nations Covenant. His Department is also one of the nations that he is called upon to consider from time to time.

It will be evident that when responsibility is so scattered that there is little hope for a demonstrated point of view as to championing a National Air Policy. With this in mind, the second demonstration in the proposed National Air Policy is that it is broken, charged with championing a national air policy, is needed by the Government. This would not necessarily require a new effort in the governmental organization. There are probably too many already. But there is a need for more civilian air men who the President could rely for advice and independent consideration of aircraft problems. The National Advisory Committee for Aeronautics has often acted in this role, but composed as it is mainly of the heads of air services of the government, it cannot be expected to give independent advice.

Just over time, therefore, as aviation takes its place as one of the important activities of the government, the best suggestion that can be made is that of charging some civilian air person in the government with the duty of championing a National Air Policy.

### Chasing the Records

**T**HIS of the driven new American airplane records which have just been established by the Coast Committee of the N.A.A. caused the general world records of the corresponding classes. As a result, the N.A.A. has requested the International Aeronautic Federation to recognize these records as official world records.

Leslie J. A. Macready and Harold G. Gatty, A.S., and Leslie F. W. Wood and F. D. Price, U.S.N., who made this possible lot of records have been furnished named proof of American determination not to let any world record go

out of the United States, and of recognizing those that may be held by any other country. What adds to the value of this demonstration is that with one exception everyone of these records was made in an American built aircraft equipped with an American engine. The Army T-2E plane, fitted with the Liberty engine, and the Navy G-2A biplane, built by the Curtiss Aeroplane & Motor Co. and fitted with the Wright T-3 engine, all share the glory of this new achievement.

It is understood on good authority that from now on all record attempts which the Army Air Service may undertake will be made in all-American aircraft, that is, machines designed as well as built in this country, and equipped with American designed and built engines. In the Naval Air Service all record records now to be made owing to the virtual abandonment of war time equipment.

### Will He or Won't He?

**F**REDERICK B. PATTERSON, president of the National Aeronautic Association of U.S.A., announced at a dinner tendered last in New York upon his return from Europe that Sadi Lecoq, the famous French pilot, would be entered in the Pulitzer Trophy race to be held at Dayton, Ohio, next October, and that he will fly a Devoianer motor now being built for this purpose.

An answer reproduced this statement at the time, our readers will be interested to learn that *L'Auto-Sports* of Paris in its issue of June 4, 1923, recently stated that Sadi Lecoq, if he will participate at all in the forthcoming Pulitzer Trophy race, will fly any plane other than one of the Napier-Devoianer company, of which firm he is the chief test pilot.

In view of the fact that the National Aeronautic Association announced last year before the St. Louis Air Show (see Aviation, Oct. 1, 1923) that the Italian pilots Silvio Pavesi and Mario Pavesi would participate therein, which neither of them did as a matter of fact, it would seem doubtful that the N.A.A. should the Sadi Lecoq's entry at an early date, rather than wait for the one of the time to do so.

Further advice from Paris indicates a very great uncertainty as to any French entries. It appears that the representatives there have very wisely decided to determine whether their machines have any chance at winning before making their registration as a so called "international" race. Calling these more international just because foreign entries are invited seems to us to cover a title of defining the public. At Detroit and at St. Louis there was created an impression that this event was an international affair whereas it was merely a series of races between our Air Services. Now, with the Navy out of the room it would be much nearer the truth to call it an Army Air Service meet than any other. Correspondents on such matters, where the War Department is participating in such a degree would seem to be a matter of plus late during the public.

# Progress of American Round the World Flight

Lieutenants Smith, Wade and Nelson Have Completed 18,450 Miles  
Out of 24,000 Called for in Schedule



When, on July 17, the three Air Service World Cruisers sailed from England, they completed 18,450 mi. out of 24,000 mi. called for in the schedule for their great feat. They still have a distance of about 6000 mi. to fly before they reach San Francisco, Calif., their point of initial departure from the United States. As Stuart Madsen, the British world flyer who is circling the world, the other way, has still about 11,000 mi. to go, and the worst part of the trip is now. There is little chance that he can beat the American public before reaching home, and, anyway, he cannot then from completing their trip on scheduled time, five men in.

## The Ships and their Crews

In having are the names of the three American round-the-world planes and of their crews.

**Plane No. 1—Pilot, Lieut. Lloyd H. Smith, M. S. M. A. 1st Lieut. Lester P. Arnold, M. S. M. A. 2nd Lieut. Henry B. O'Brien.**

**Plane No. 2—Pilot, Lieut. Lloyd H. Smith, M. S. M. A. 1st Lieut. Lester P. Arnold, M. S. M. A. 2nd Lieut. Henry B. O'Brien.**

**Plane No. 3—Pilot, Lieut. Lloyd H. Smith, M. S. M. A. 1st Lieut. Lester P. Arnold, M. S. M. A. 2nd Lieut. Henry B. O'Brien.**

The three planes were originally designated with the numbers 1, 2 and 3. Plane No. 1, the Seattle, was May P. L. Martin's ship, which was lost in Alaska.

One world flyer, one due to complete their long flight in June. Madsen, Calif., on about Sept. 3, if all goes well, minutes of the Army Air Service indicate. This is hard question steps in Hall, England, for the change from ships to airplanes and radio instruments, and several other steps in the question of bad weather.

The estimates are based upon the planes reaching London on July 15, and staying there ten days before the final leg on the north Atlantic islands is undertaken. They should reach the mainland of North America, Labrador about Aug. 15, then flying to Newfoundland, Nova Scotia, Boston, New York, Washington, and then the transcontinental jump, for which three days may be allowed.

Following is a brief log of the American World Flight, giving the time elapsed to cover the various legs, and the principal incidents.

## Log of the Flight

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# Fittings and other Structural Airplane Parts \*

By P. EYDAM

The strength and reliability of airplanes depend greatly on the careful design and construction of the fittings, couplings and other highly stressed parts. In designing these parts, attention must be paid to the possibility of supplementary stresses, which in certain cases are experienced only by alterations in force during flight or during the landing tests.

Some fittings are made with the smallest possible dimensions, for the purpose of saving weight, and unnecessary welding is extremely explicated, work on these parts must be carefully supervised, in order that they may not be weakened at possibly

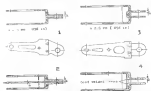


Fig. 1 Wing spar joint. Fig. 2 Same reinforced. Fig. 3 Corresponding fuselage fitting. Fig. 4 Same reinforced.

welded joints by badly formed welds, nor the material harmed in the neighborhood of the welded areas.

Failures of fittings, which occur during strength tests, are due in part to faulty design, but more often to careless workmanship. Hence the use of unnecessary welding cannot be done in the construction of fittings, and the use of welding is, in fact, the possible weakening of stressed sections should be carefully avoided, and it is of special importance that welded areas should not be permitted to carry the load alone.

The possibility of this is evident from the fact that a fastener having a tensile strength of 10,000 lb. may be used for a fastener which is not used for a new one of improved design, or cooperation between the fitting station and the fastener, both attention, however, necessitates the expenditure of money, time and material by both parties, which earlier can usually be avoided by the exercise of sufficient foresight. In many cases, these tests produce an structural change and ruptures, whereas alternative stresses, vibrations and the influence of the weather produce new conditions that rupture sections, although the parts have been carefully designed. The most important parts must, therefore, even when of proved strength, be continuously tested for increased loads and also in order to reveal the possibility of defects causing in stress subsequent quantity production.

The following remarks relate to actual examples, which have been observed in strength tests on airplanes.

## Fittings for Wing Spar Joints

A case (Fig. 1) under showing a non-reinforced joint at 90 per cent of the required load, after undergoing strength tests without noticeable alterations in form. The structure showed numerous flaws. During the non-destructive test, the legs around the wing to the fuselage were subjected to a strong bending stress which proved too great for the welds at the base of the

legs and the latter were weakened off. The short metal was also bent sharply back and broke with the leg. It was found that the parts were only superficially connected by the welds, the same being only 1 in. (25.4 mm) thick, as against a plate thickness of 2 mm. (0.079 in.).

This fitting was, therefore, altered (Fig. 2) by using instead of the fuselage spacing piece, an intermediate eye, which was held in place by the bolt through the spar, the bearing the weld. This strip was a little thicker than the fuselage piece, while the legs were connected strongly. With this method of construction, even an imperfect weld will not necessarily cause premature rupture. The original had weighed 135 g. (4.75 oz.) while the altered one weighed 1 g. (15.4 oz.) more.

In the corresponding fuselage fitting (Fig. 3) the legs were also strengthened (Fig. 4), the ends of these, in the rear of the fitting, being bent away from each other so as to facilitate location for landing stress in the interior of the shoe and the welds. The weight of this fitting was increased from 133 to 168 g. (4.73 to 7.0 oz.).

In a coupling between an upper wing spar and the lower (Fig. 5), the end was bent during a gliding flight test and the eye-bolt fitting was subjected to a heavy bending stress toward one side. If the end of the shoe is cut back, the upper half of the fitting can be easily repaired.

In designing this part it was properly assumed that the load on the main bearing wires transmitted to the spar would be sufficient to counteract the bending moment from the load at the end position of the fitting. In order to reduce the load-

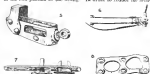


Fig. 5 Coupling between upper wing spar and cable. Fig. 6 Tension spar connection. Fig. 7 and 8 Deformed eye plate fittings.

ing stress, the eye was brought nearer to the base, and the transition from the spar to the eye was made more gradual, thus reducing the shearing stress near the eye. Although tension tests must be avoided in all such fittings, in every way to the weld the eye-bolt all around in the shoe and not to bend the wire to one or two parts.

It is frequently the case that the bolts load between the spar parts of the lower wing, is not sufficiently taken up by the transverse members of the fuselage. Transverse bars or compression tubes, which at the same time take up the load when landing, are, therefore, recommended for use with such fittings. These must, however, be as connected to the fittings, so that the spar is coupled as to achieve the possibility of any bending stress being set up by the attachment. Often the transverse members did not lie in the plane of the stress, so that rupture resulted through the supports after landing maneuvers.

The tension member in Fig. 6 shows a defect of the fitting. This member consists of slugs secured to the fuselage, so as to give additional stiffening, and strips being connected to the fuselage fittings by bolts. Since these bolts lie under the

July 25, 1934

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position of tension, bending moments arise. The legs, which the bolts pass, bend and the bolts break. The bolts either be connected so as to bring the bolts into the plane or they must be replaced by wires with tension.

## Fittings for Strut Connections

Eye-bolt fittings have frequently failed, both under test and in flight, the fitting usually being at the hole in the plate (Figs. 7 and 8). Owing to the oblique direction in which the tension in the wire acts, one edge of the eye-plate base is pushed into the work, while the other is up the flange plate, the eye-plate thereby shearing off. This is chiefly because the transition from the base of the eye-plate to the leg is abrupt. These eye-plate fittings are still unsatisfactory for taking up loads in an oblique direction and it would be better to dispense with them entirely in the future.

It is difficult to ascertain the forces acting in these fittings, and for the purpose of determining suitable sizes for eye-plates and legs were tested with two different types (Figs. 9 and 10). Each fitting was attached to a steel plate and subjected to a load  $P$  acting at an angle  $\alpha$  to the axis. The tests showed that with an eye-plate of 25 mm. (1 in.) base diameter and 5 mm. (0.079 in.) thick (the legs began at 1690 lb. (7500 lb.) and the eye-plate shear through at 1600 lb. (3500 lb.). When the flange was 16 mm. (5/8 in.) thick, the rupture first began at 1600 lb. (3500 lb.) and the rupture first began at 1600 lb. (3500 lb.) and the eye-plate was the first to give way. Tests with eye-plates of 25 mm. base gave no better results. The last tests showed that the breaking load did not increase in proportion to the thickness of the plate, but remained comparatively small for the thicker plates.

The connections were improved by increasing the thickness of the legs from 1.5 to 2 mm. (0.059 to 0.079 in.) to 2.5 mm. (0.098 in.), for comparison being made the same thickness.

In attaching the fittings to the spars, the plates are usually set into the spar, or into the plywood with which it is covered (Fig. 11). The method depends too much on careful workmanship and of less value happened that the base was not accurately fitted so that the plate was bent while being secured into place. It is, therefore, better to place a specially shaped metal nut in under the plate.

In the next fitting (Fig. 12), the bending moments were advantageously reduced by placing the bolts in sockets which were pressed out of the plate, comparatively deep. The spar

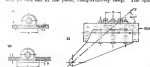


Fig. 9 and 10 Different types of eye plate fittings. Fig. 11 Corresponding spar attachment.

was bent through at the central axis and completely covered by the fitting, so that the forces were uniformly distributed throughout the fitting. Such a fitting was tested under a load of 1600 lb. (3500 lb.) and the bolts passed, in order to prevent the latter being bent. For this reason, the bolts were somewhat longer. The fitting weighs 560 g. (12.75 oz.), but can be improved by making the eye-plate thinner.

In a strut fitting with shackle and wire attachments (Fig. 13), the eye of the strap wire was weakened off during the test by a very low load resistance. The eye did not lie in the direction of the tension, it was cut out of it, but the eye was subjected to excessive stress. Should the strap wire be attached at another point, this connection could be recom-

mended as being suitable and light. In many cases, the shackle bolts were bent (Fig. 14) thus causing rupture.

The actual construction of shackles (Fig. 15) is capable of much improvement. When the shackle plate is cut out at the middle of the bolt (133a), the latter now carries nearly 1/3 the load. The stress shown in 133b, owing to a further decrease of the bending moment, can carry almost double the load.

The best form of shackle, shown in Fig. 15d can take 4 1/2 times the load of an ordinary shackle (Fig. 15b), if it is made equal to 0.05 in. Shackles of this type have broken

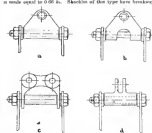


Fig. 13 Different types of shackle design.

strength up to 3500 lb. (7800 lb.). In this case, the shackle plate and the bolt begin to shear, as can be seen in the illustrations on the left of Fig. 14. A load of 1500 lb. (3300 lb.) was acting in the opposite direction. In no case was deformation of the shackle itself observed. In a shackle of the old type (see right of Fig. 14), the bolt bent under a load of only 1500 lb. (3300 lb.).

In the alternative strut attachment for all-metal airplanes (Fig. 16), owing to insufficient play between the strut and the fast, the former is subjected to a bending stress when the wing is bent, which the result that the strut eye must, therefore, be checked.

The Bendley-Pope airplane shows carefully designed fittings. Every connection was thoroughly checked, being attached to the spars by bolts through the neutral axis.

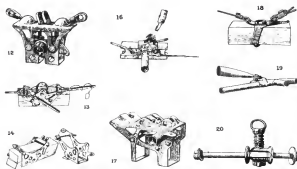
In the light and efficient fitting shown in Fig. 17, attention must be paid to the fraying at the bearing wires legs, to prevent that they are not too slack. In several cases, the legs were so short as to prevent the bolts from fitting properly, with the result that they were subjected to bending stresses.

## Internal Bracing

Since the internal bracing of the wings is subjected to heavy stresses due to bend resistance, the rigidity of the wing can be greatly increased by the careful construction of this bracing in an interior, by distributing the area thereof, increasing the section of the string in contact with the spar and thereby strengthening the spars, the strength to meet bend loads was suitably made adequate. Defects in this direction must, however, be discovered and corrected as far as possible by giving constant tests or strength calculations, in order that serious alterations may not be necessary after the delivery tests.

One fitting (Fig. 18) was bent during the test for bend resistance, owing to insufficient width of the strap. The width of the strap at the point of contact with the spar should be at least equal to half the circumference of the spar. The equivalent increase in weight is more than offset by the re-

\* From Technical Review, Vol. 10, No. 4, pp. 199-208 (1934) S.A.E. Technical No. 110.



Figs. 12-13-14. Different types of wire fittings. Fig. 15. Stent attachment for all-metal airplane. Fig. 16. A life and offset fitting. Fig. 17. A fitting that failed. Fig. 18. Stent fastening between shifter and pin. Fig. 19. Hinge pin attached to a roller in a pin.

excess stiffness of the wing. The eyes of the legs for the external bracing frequently break off and need subsequent strengthening.

#### Bracing Wires

The fracture of another variable being frequently happens, coming to the extremely severe cases being badly tapped, thus producing bending stresses in the stressed ends, in addition to the tension for which they are designed. In one case, the steel transverse the stress was so seriously tapped, that the eye of the wire was bent out of center by a distance equal to the diameter of the wire. Under these circumstances, the additional stress in the place of the eye amounted to an increase of 80 per cent in the breaking load on the stressed end. As the stress on the edges of the eye is still greater, failure at this point is easily explained, especially if the parts are brittle. The severity can be diminished by leading the stressed ends, but such wires can frequently be broken in this way. In particular with rounded eyes, fitting between forks, the pin is subjected to bending stress, though it should not be subjected to bending stress.

Continuously made spheres are a frequent cause of trouble. Since it is very easy to slip the outer layers of the cables, and other wires, it also into the central portion. Although tests have shown that a sphere can prevent sufficient strength when the outer layers are slipped, the safety of the sphere as a wire depends on the stress exerted by the surface. In such sphere, even under load tests, the case being pulled out of the surrounding sphere layer. If the sphere is carried through in the core, defective workmanship occurs more rarely and can be quite easily discovered by external inspection.

Spheres often break because they were soldered, not with a soldering iron, as provided, but with the aid of a blow lamp. The wire being thus softened. The necessity of using a soldering iron is also frequently overlooked. The liquid then remains between the wires, which in time corrodes and has strength.

Over to complaints from the front, a number of splices in use lately were noted. While the percentage loss in strength at the splice is 16 per cent of the strength of the cable, the strength of the splice investigated varied between 50 per cent and 53.4 per cent of the strength of the cable. Such splices are extremely quite coefficient. It was also found that hydrostatics, and was used in soldering the splices, and that many of the related joints had corroded. This one shows that splicing needs especially careful supervision and that breaking tests on selected splices are absolutely necessary.

#### Control Gear

A common form of strut (Fig. 18), which serve to connect the horizontal bracing with the fuselage under test. The fitting of the fit and handle under a very small load, thus subjecting the strut to a bending stress, which can be avoided by the use of double grommets, or, as is often specified by, may vary between the connector legs and the fitting. Such conditions are also dangerous, in view of the possibility of their being bent during the transportation of the airplane on ground vehicles and subsequently breaking under the normal load which they are supposed to carry during flight.

The hinge joint attaching the roller to the fit (Fig. 19) can be recommended as safe and practical. It weighs only 0.75 lb. (3.41 lb.). Frequently these joints are made with only two eyes. In such a case, however, the joint is subjected to a bending stress and, with large roller loads, such joints have a tendency to jam. Since these large joints also make it very difficult to remove and replace the rollers, it is desirable that they should be employed more frequently for attaching the rollers.

The structure of the rollers in frequently too weak. Since they are subjected to alternating loads and vibrations during flight, these joints should be particularly rigid. For this reason, it is also desirable to make the rollers, between the front roller and the trailing edge of the airframe, of low form. Thus the often observed serious failure would be avoided.

## Aeronautical Board is Reorganized

Secretaries of War and Navy Agree on New Functions of Board

The latest command in AVIATION on the newly adopted plan of the War and Navy Departments regarding the Aeronautical Board was based on administrative changes. No that one random war form, a complete draft of the new plan, the official text of the order is printed below in full.

The following order supersedes the existing precept of the Aeronautical Board:

#### To Prevent Duplication

"In order to prevent duplication of effort and to secure a uniform measure of cooperation and coordination in the design of and employment of the Army Air Service and Naval Aviation, the Secretary of War and the Secretary of the Navy, agreed upon the reorganization of the Aeronautical Board with functions, membership, and procedure as follows:

1. The Aeronautical Board will investigate, study, and report upon all questions affecting jointly the development and employment of the Army Air Service and Naval Aviation, referred to it by the Secretary of War, by the Secretary of the Navy, or by the Joint Board. It will also be kept with its duty of engaging consideration of such subjects when, in its judgment, it is necessary, and of recommending advice it considers essential to establish cooperation and efficiency of cooperation and coordination of effort between the Army Air Service and Naval Aviation.

2. In the execution of the foregoing, the following will prevail:

(a) All recommendations of the Aeronautical Board affecting joint policies or joint plans for the tactical or operational employment of aircraft or for the location of air stations, will be addressed to the Joint Board for consideration and recommendation to the Secretary of War and the Secretary of the Navy.

(b) The Aeronautical Board will formulate and submit to the Joint Board for approval, suitable joint Army and Navy aircraft, or other matters, which are referred to it by the Secretary of War or the Secretary of the Navy.

(c) The Chief of Air Service of the Army and the Chief of the Bureau of Aeronautics of the Navy will submit to the Aeronautical Board all questions which concern jointly the Army Air Service and Naval Aviation.

#### To Control Development Work

(a) The development of new types of aircraft, aircraft motors, and aircraft accessories, or of equipment to be used from aircraft, or of any other matter, which is referred to the Army or Navy and shall be carried on only by the Service to which assigned. This instruction will not prevent the employment by either the Army or Navy of any type of aircraft or of any other matter, after development are considered to be necessary for the accomplishment of its functions. Questions relating to the development of new types of aircraft, aircraft motors, and accessories, or of equipment to be used from aircraft, referred to the Aeronautical Board for recommendation as to whether the Army or Navy shall be charged with the development.

(b) Whenever possible, training, repair, and other aviation facilities of either the Army or Navy will be made available for use by the other Service.

(c) The Aeronautical Board is specifically charged with the following:

(a) Plans to prevent competition in the procurement of aircraft. Before arranging to purchase aircraft, each Service will ascertain whether aircraft of the type desired are in demand from the other Service, and if so, the procurement of materials in case of war will be submitted to the Joint Aeronautical Board.

(b) Coordination and recommendation of all projects for experimental stations on sites, for aerial air

stations, and for stations to be used jointly by the Army and Navy, or for extensive additions thereto.

(c) Consideration of and recommendation in regard to all matters of appropriation for the aeronautical programs of the Army and Navy before such estimates are submitted to Congress.

(d) Coordination of the activities of the Army Air Service and Naval Aviation in the pursuit of common aims and coordination of their activities with the aeronautical activities of other branches of the Government and with the civilian aeronautical organizations.

(e) Recommendation in order to secure co-operation or invitation of foreign governments to participate in aviation activities abroad and on the activities of civilian organizations to participate in such activities in the United States.

(f) Compliance of the issuing of licenses to vehicles in appropriate aircraft in time of war.

#### Membership of Board

"1. The membership of the Aeronautical Board will consist of:

(a) For the Army: The Chief of the Air Service, the Chief of Training and War Plans Division, Officer, Chief of Air Service, one member of the War Plans Division, General Staff, War Department, to be designated by the Chief of Staff of the Army. The last named officer, while holding such office, will not be eligible for duty with the Joint Board or Joint Planning Committee.

(b) For the Navy: The Chief of the Bureau of Aeronautics, the Chief of Planning Division, Bureau of Aeronautics, one member of the War Plans Division, Office of the Chief of Naval Operations. The last named member while holding such office, will not be eligible for duty with the Joint Board or Joint Planning Committee.

"3. The Secretary of the Board will be selected by the Secretary of War and the Secretary of the Navy, from the principal personnel of either Department and detailed to this duty.

"4. The Aeronautical Board will hold regular monthly sessions and such extraordinary meetings as may be deemed advisable. The Board will prescribe its procedure. The session number of the Board period will precede all its reports. The reports and correspondence of the Board will be authorized by the Senior Officer of the Army and the Senior Officer of the Navy present at the meeting at which the action was taken."

(Signed.) JOHN W. WALKER

Secretary of War.

CYRIL H. BRYAN

Secretary of the Navy.

By action of the Secretary of War and the Secretary of the Navy, James D. Hatcher, the Secretary of the Joint Board has been appointed Secretary of the Aeronautical Board.

#### New World Duration Record

A new world duration record without refueling was made on July 16-17, 1919, when the French aviators Compt and Brochenet completed a flight in a Fokker plane 37 in 36 hours and over a circuit laid out between Elzevier and Chartres, France.

The new performance bettered by 3 hr. 54 min. 36 sec. the unrefueled world duration record made by Louis John A. Mearns and Oakley G. Kelly, A. S. on a Fokker T2, run by 40 hrs. 44 min. 44 sec. the refueled world duration record made by Lewis, Lowell H. Smith, leader of the American World Flight, and John Kieffer as a 101HR.

### Airplane Deafness

In an article written by Capt. Verner T. Scott, Flight Surgeon at Wheeler Field, H. T., on the above subject, he states that airplane deafness is a temporary condition caused by the vibrations of an airplane under and above the flight line of our best's duration or longer and later from one to several hours, depending upon the length of the flight. That airplane deafness is a temporary condition is due to two factors—the compression shortness of exposure to the motor vibrations and to the partial protection afforded the auditory apparatus by the helmet. An aviator who recently flew across the continent in thirteen hours stated after leaving one of seven hours for deafness and "ringing in the ears" continued all night and sometimes until noon next day. In the average

motor but allow sufficient noise through to determine a rising motor. Altitude and the rush of air prevent the powder puff from becoming uncomfortably warm in summer. One should be taken to select a powder puff of wool or a mixture of wool and cotton, as the cotton alone becomes very soft after a little use, whereas the wool fibers stand out and remain soft and fluffy.

Commenting on Captain Scott's article, the *Los Angeles Times* remarks that the plan and states that it was rather generally followed on our border patrols and found very satisfactory by the pilots. It is believed by the flight men that most of the distressing symptoms from which the pilot suffers are due to the engine in lead down by Captain Scott's observation, that "air vibrations" caused by the motor, the propeller and the rush of the ship through the air are all responsible



Where metal screws—Carburized duralumin propeller as it looked after an airplane accident (lower picture), and after it had been reconstructed by the manufacturer (upper picture).

cross-sections, first of two kinds, the deafness will last from one half hour to several hours. The deafness is not a permanent condition; it is intelligible and conversation can be carried on only in very loud tones.

It is believed by some not familiar with aviation deafness that the deafness is due to the change in atmospheric pressure rather than the noise of the motor. That atmospheric pressure is not a factor in producing airplane deafness is demonstrated daily at the School of Aviation Medicine, Maxwell Field, Long Beach, H. T. This school has for training and research purposes a large "low pressure chamber" in which the barometric pressure can be lowered from 760 mm. (sea level) to 560 mm. (equivalent to 30,000 ft. above sea level), and although deafness is not a factor in very deep diving, the experiments there have never been a case of deafness experienced by anyone "going up" in this chamber.

It is changes in atmospheric pressure when a factor in producing deafness, it would never prove really true. It is actually flight because of the more rapid change in pressure. To gain the same rapid change of atmospheric pressure in actual flight an aviator would be compelled to ascend at the rate of 1,200 ft. per minute to 30,000 ft. and descend at the same rate. That, of course, would be done only in calibration flying.

One would hardly consider a lady's powder puff as a means of protection for airplane deafness, but that is exactly what Captain Scott recommended. He stated that the simplest, safest and most practical arrangement is that made by sewing the powder puff on the inside of the ear flaps of the helmet, so that the puff fits snugly over the ear. It is very much like a case being behind the ear. The powder puff will cushion out along the sides of the face at the first blast of the propeller and will not only muffle the noise but will catch all the dust and dirt.

The use of ear plugs made of hard rubber, paraffin, or wax should be discontinued. They not only have a tendency to irritate the auditory canal in any other foreign body, but they tend to become soft and become hard. Another objection is that, being soft, they are likely to become hard when exposed to heat.

Cotton plugs find favor with some fliers, but their use is not recommended, either there and remains subject to the usual, common criticism and deafness will be renewed by a physician.

The powder puff is the best of all means of protection, and, especially, cannot possibly irritate the auditory canal, and are always with the helmet. They block the loud explosion of the

motor for the increasing number of cases of partial deafness. Long years spent in the air caused great damage for the pilot, pressure deafness, temporary loss of vision, and loss of hearing, and being not any latest physical defect. These symptoms are not so pronounced in the front cockpit as they are in the rear cockpit. The deafness is not so pronounced and appears to be a special ailment in the long sea pressure zone in the rear cockpit disability. Therefore, it is to the pilot's self interest to protect himself in every way possible.

### Metal Propeller Scores

Some time ago a Navy test plane was wrecked at Pensacola, Fla., when the pilot saved the machine on landing. The ship was equipped with a Carburized metal propeller, which is a duralumin forming, and that was very badly bent at the tip, as shown in one of the accompanying illustrations. It was first intended for the dump heap, where someone, with typical Navy economy in mind, suggested that it might be strengthened. Accordingly it was returned to the manufacturing firm for remodeling, and when successfully accomplished, as will be seen from the other illustration, for the strengthening cut process the propeller was remodeled, and after it had been strengthened, treated and balanced, it was found to be as good and useful as the original one.

Metal has long been considered to be superior to wood for airplane propellers, steel being most commonly employed, but it was not until forged duralumin was used that real progress was made. Duralumin weighs approximately one third as much as steel with the equivalent strength. Its ability to withstand heat makes duralumin propellers the propellers of the future, as they are light, powerful, light green and cold in appearance, and are important. Duralumin propellers are at their best in air or tropical climates where the use of wooden propellers is a constant source of annoyance, but where metal propellers are in use they are at their best.

Due to the greater strength of duralumin over steel for a given area, it is possible to use thin blade sections which are much more efficient than the usual thick blades of wood. The Carburized metal propeller is a very strong material, and its use in a metal propeller was very strikingly shown in tests made with both types of propellers on last year's Pulitzer prize. The machine when equipped with a metal propeller was found to be in better condition than when equipped with a wooden propeller.

### Plans for Schneider Cup Race

The general plans for holding the Schneider Cup Race are as follows: The best of the race will be held at the end of an important event such as the date of the race, situated about 100 miles from Baltimore, in communication through by the electric lines across and good roads.

The plan of the race and aerial photograph will show the general layout of the park. The course of the race is about an isosceles triangle of 30 miles, resistance being required to make some laps.

In the details of taking care of planes and personnel of the various contestants, it will be noticed from the photograph that there is a white sandy beach to the right of the small pier. Here will be erected 12 one-plane hangars with a total area of 100 ft. square, communicating with a 100 ft. square which will slope down the sea wall down into the water, ending at each a depth as to give the planes flotation when mounted on their trailers. The hangars will be provided with electric lights, water, and other facilities. There will be provided for parking the planes up the runway. A car and work shop each as used in the Army Air Service, including a table, drill press, motor wheel and complete set of tools, will be provided on the pier. During the race each plane will have assigned a power boat for towing purposes.

The flying and mechanical personnel will be located as follows: The Flying Club of Baltimore will provide rooms at the Southern Hotel in Baltimore for racing season this year. The mechanics personnel will be located in the building above on the pier between the fountain in the circle and the pier lower. This is the restaurant building of the park. For each contestant will be on the second floor where are quiet facilities and a large sleeping room which will be provided with army cots and blankets, and six single rooms which will be made available to the team captains. First and last is provided. On the first floor the dining room will be arranged and immediately adjoining there is a large kitchen capable of supplying many times the number we will have on hand. Here, the Flying Club of Baltimore will take care of the team, per gallon, plus one crew captain, crew member, fuel and all will be provided by the Flying Club of Baltimore. In purchasing the fuel, the supplying firm will comply with U. S. Army Air Service specifications as to fuel and oil.



Airline view of the opening race selected at Baltimore for the forthcoming Schneider Cup race

It is suggested that those persons who may desire to see the Flying Club of Baltimore of the season that they will and, so that they can be provided.

As to shipping arrangements, it is advisable that contestants should arrange to ship direct to Baltimore. The following information is available regarding shipping lines to Baltimore. The shipping representatives have advised that since the end of the Atlantic Transport Line of London should be the Atlantic Transport runs regular sailings direct to



Course of the Schneider Cup race to be from Oct 24-25, next, at Baltimore, Md., and the American defender team

Baltimore and we will provide means for landing the planes on the pier and moving them direct to the house of the race. From Italy, the Riviera, Trieste, and San Francisco, Trieste, Italy, should be used. This line, however, is subjected to heavy epidemic sickness and it would be well to have foreign representation on board with this line as to whether they will have a ship coming to Baltimore about the proper time. It is to be pointed out that shipment to New York will require transshipment by rail to Baltimore, all becoming complicated.

It is suggested that all material direct be consigned to the firm of William H. Moore, Stewart, Baltimore, Steamship Agent and Custom House Broker, Baltimore, Md., and that copies of bills of lading and consular invoices be obtained, sending them forward by the fastest means, so that the firm may prepare the necessary entries and bonds for calibration points to avoid any delay upon arrival at Baltimore or New York, as the case may be.

### Book Review

THE COMPASS AIRMAN. By G. C. Bucky, D.S.O., R.A.F. 259 pp., numerous illustrations. (E. P. Dutton & Co., New York.)

In this comprehensive volume the author provides the airman with a complete outline of the knowledge he ought to possess, the mechanical and physical theory of flight, particularly of manufacture which refers also to the construction of the plane, the actual design of the different types and their respective parts, including the engine and all its subsidiary details, the fuselage, the control system, the rigging and the propeller, the action of air streaming and running the engine and all the instruments required in flight, as well as general instructions for preflight aerial work. Finally the various buildings of an airframe, and chapters on the method of keeping the machine in condition and some useful meteorological data.

The author's style is simple and his manner of presentation clear. Where needed, designs and mathematical formulas are given, and care has been taken not to overload the text with detail either unnecessary or too creditable.









fresh green fields, and clusters of bright lavender willows emerge into the gold and green beds of meadows and meadow grass.

## U. S. NAVAL AVIATION

### New Launching Catapult

A new type of airplane catapult in which a powder charge is used for giving the plane its initial start instead of being powered up has recently been tested by the United States Navy with satisfactory results.

In the new type the catapult gun contains a piston, which is connected through a series of multiple pistons or rollers to a small wheeled car. The car, on which the airplane is placed, runs on tracks, which are 30 ft. long and are secured to a platform located on the top of a battleship turret. When the powder charge is fired in the gun, the piston being forced to move, sets through the pistons, thus pushing the car rapidly forward along the tracks at a speed of forty miles an hour, carrying the plane with it, over the Kansas City Star.

When the car reaches the end of the track it is stopped by

means of hydraulic and spring buffers. The plane, with its engine going full speed, is automatically released from the car and continues under its own power.

In the test made at the naval air station at Washington, a single motor mount airplane was used. The plane left the catapult car and continued its flight without any drop in altitude below the level of the tracks, which is unusual. The test was particularly pleasing to the officials, since hitherto it has been necessary to leave the tracks at such a height above the water that catapults were impracticable for use in the shape of an aircraft carrier.

The principal advantage of the new type catapult over the compressed air type lies in the speed and facility, with which a number of planes may be catapulted from a given carrier. In using compressed air it is necessary to reduce the air to its proper condition, which takes considerable time. By the use of the gun type the enormous space and weight of the air compressing machinery will be eliminated. All airplanes carry and other types of ships carrying airplanes will be equipped with the gun type catapult as soon as the Navy Department can build and install them.

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**SAN ANTONIO AVIATION & MOTOR SCHOOL**  
Flying instruction and aircraft repair.  
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## A Suggested National Air Policy

That a National Aviation Policy is needed by the United States is obvious. To get such a policy in concrete form AVIATION requested several thoughtful friends of aeronautical progress to make suggestive and constructive recommendations. Some of them are given below and will be printed each week with additions, omissions and such other changes as appear to be helpful toward the formulation of a sound national air policy. Readers of AVIATION and others can render no greater service to the cause of aeronautical progress than contributing their comments and suggestions.

### GOVERNMENTAL.

A continuing program of aircraft development both governmental and commercial.  
A civilian, charged with championing a national air policy, is needed in the Government.  
Aircraft construction in the House and Senate to build aircraft hearings where civilians as well as government officials can express their opinions.  
A detailed annual budget for all Governmental Departments, and an annual statement of all expenditures.  
An experienced staff of flying officers at the head of all governmental air defense services.  
Coordination of all procurement and experimental aircraft work of the government under one agency.  
Limitation of government manufacture to repair of aircraft and specialized work that cannot be done by private firms.  
The elimination of the duplication of aerial functions and facilities by government departments.  
A country wide Air Mail system of trunk lines connecting the principal cities of the country.  
Establishment of a National Airway System through cooperation of the Federal Government with States and Cities.  
A national aircraft law that will regulate aviation, administered by practical pilots and experienced aeronautical engineers.  
Membership of the United States in the International Convention for Air Navigation.

### COMMERCIAL AIRCRAFT OPERATION.

Creation of commercial air lines by private enterprise or government subsidy.  
Encouragement of participation by private companies in aircraft races and competitions.  
Encouragement of the training of pilots by civilian schools.  
Creating an Exposit de Coupe around flying men all over the country by frequent gatherings at aviation meets.

### INDUSTRIAL AIRCRAFT CONSTRUCTION.

Recognition that a sound aeronautical industry is a prime necessity of our National Defense.  
An active industrial association that will coordinate the aircraft industry and defend it from attack.  
Encouragement of the designing of new types of aircraft by manufacturers by allowing them to retain their proprietary rights.  
Concentration of manufacturing firms on specialized types of army and navy aircraft.  
Encouragement of research by constructors, universities and other agencies as well as by the government.  
Encouragement of an annual design competition for commercial aircraft.

### CIVILIAN.

A national aeronautical organization composed of public spirited citizens that will take a strong position of leadership on national aeronautical policy.  
An Annual Aviation Week during which the country will think of aerial progress.  
The formation of local aero clubs by firms for the purpose of stimulating flying in all localities.  
Encouraging the public to fly and patronize the air mail and transport facilities.

## PUBLISHER'S NEWS LETTER

There have been several requests from readers of AVIATION for more detailed information regarding the progress of the Round the World Flight. So that the reason for the scarcity of news concerning this flight may be known, we quote from a letter received from the office of the Chief of Air Service:

"The data which you have in copy of log cut is as about as complete as we have ourselves. The fleet have not been encountered with making out long detailed reports at every stop, which has produced one reason for the scarcity of the Round the World Flight, thus far. The lack of arrival and departure is about the only information we receive from the flight commander."

Until we were sure that further information was not available we have not published a log of the flight thus far. With this issue we commence the publication of a log of the flight from the beginning. It is largely a compilation of newspaper dispatches and official bulletins from Washington. Some of our readers have written us that as there have been rumors in their local newspapers that the flight AVIATION might never actually occur. This we are attempting to do, and for the remainder of the flight this log will be added to each week until the flight is completed.

We are often asked "What is the nature with aviation in this country?" There are many considerations and answers made that apply to special fields. It is consequently very to make constructive suggestions as to the governmental regulations but in fairness it should be realized that the officials in charge are limited in their action by law, by precedent, by antagonistic associates in the departments and in some cases by lack of experience. Critics of the progress of those who believe in the most publicity given them from their connection with aeronautic activities also reveal many judgments and petty troubles that tend to hold back real progress. The industry that has been struggling to hang on until the dawn of the aerial day, has had quite enough criticism, most of it coming from disappointed sources, or prompted by self-interest. In fact self-interest is so controlling in almost all that happens in aeronautic history that it might as well be called by its real name.

From its point of view, AVIATION can finally say that it believes that one of the shortcomings of the aeronautical industry is a lack of foresight in establishing that most essential of all business assets—goodwill. With very few exceptions the aircraft manufacturers go to the principle that they cannot afford to create goodwill until they get orders. And when they get orders, they find that their product will come at once goodwill. This attitude of concerning assets that instead of developing a feeling of confidence and satisfaction in aviation, the industry allows itself to be criticized and its products belittled

about by the heaves of unconfident comment. At this point it may be asked, what is the answer?

AVIATION can only point to practically every other industry that has succeeded and show that almost the first step in any budget is that of advertising. Instead of waiting for orders to produce an income part of which can be used to create goodwill—the usual procedure in the crude good-will by advertising before sales and then emphasize the merits of the product by what may be termed "follow through" advertising. Hitting the ball, to use golf language, is only the beginning of a stroke—the follow-through is equally important. Advertising, in any form, must cover the complete stroke to be satisfactory.

With these general principles in mind, it must be apparent that the aeronautical industry as a whole has been very backward in its publicity plans. It has barely relied on the mass spent and free publicity to give it prestige. While these are always factors in any plan, no amount of attention can make that kind of publicity a direct sales force. The reason for this is that it only deals with given values and not with the merits of a product. Of course, aircraft that fly around the world putify their claims to excellence as cranes, but their constructional features and ability for special military or naval uses are not emphasized.

As an illustration of the misconception that general publicity sometimes creates, a War Minister in a Central American Republic directed one of his firms to prepare to fly to the United States. When informed that it was impossible with the type of aircraft available, the Minister showed his credulity by exclaiming, "Didn't aircraft fly across the Atlantic?" It is just these general and unrealistic ideas that are created by general publicity. The only way the specific uses and constructional excellence of aircraft can be given proper treatment is through advertising prepared by the manufacturers.

It is one of the misconceptions of aviation to see men whose business interests have been based on advertising overlook this essential cog in their sales mechanism. Men of large interests who are interested in the promotion of commercial aircraft do not seem to realize that the problem of aviation promotion is just the same as any other. As we pointed out, self-interest is usually to be detected in all aeronautical discussion. AVIATION, being the only aeronautical paper that describes new aircraft in detail and gives to readers technical descriptions, justifies its belief that it is the proper place for aircraft manufacturers to tell their story to the aviation world—other publications have their own special appeal, but AVIATION is read by those who know aviation and who desire informed descriptions of aircraft and its uses.—L. D.G.

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AMERICA'S FRONTIER FLIVVER PLANE MOTORS  
ALWAYS AVAILABLE IN NEW SUPER DESIGN



12 Cyl. 16 to 20 H.P. 49 Lbs. \$275.00  
Medium Power For Power-Without Sacrificing Reliability.  
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## The Weekly Issue of AVIATION That You Miss

Because you are not a regular subscriber, may contain the article, news story, picture or advertisement which you should have read with us.

If you are a Service or a Station, let AVIATION be as an immediate adjunct to your office, because in each weekly issue is published news stories and commercial items which you cannot see in any monthly, and more important, a NEWS which appears in AVIATION.

AVIATION  
The Globe's Aeronautic  
Aircraft Magazine

The Only Aeronautic  
Aircraft Weekly

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4-CYLINDER 2000 C.U.F.

Other 1800 Type from 10 to 120 H.P. for commercial airplanes and motorboats.

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2062	Shall and scale to fit above tachometers 50 ft.	3.50
2063	Shall and scale to fit above tachometers 60 ft.	3.50
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2107	Shall and scale to fit above tachometers 500 ft.	3.50
2108	Shall and scale to fit above tachometers 510 ft.	3.50
2109	Shall and scale to fit above tachometers 520 ft.	3.50
2110	Shall and scale to fit above tachometers 530 ft.	3.50
2111	Shall and scale to fit above tachometers 540 ft.	3.50
2112	Shall and scale to fit above tachometers 550 ft.	3.50
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2145	Shall and scale to fit above tachometers 880 ft.	3.50
2146	Shall and scale to fit above tachometers 890 ft.	3.50
2147	Shall and scale to fit above tachometers 900 ft.	3.50
2148	Shall and scale to fit above tachometers 910 ft.	3.50
2149	Shall and scale to fit above tachometers 920 ft.	3.50
2150	Shall and scale to fit above tachometers 930 ft.	3.50
2151	Shall and scale to fit above tachometers 940 ft.	3.50
2152	Shall and scale to fit above tachometers 950 ft.	3.50
2153	Shall and scale to fit above tachometers 960 ft.	3.50
2154	Shall and scale to fit above tachometers 970 ft.	3.50
2155	Shall and scale to fit above tachometers 980 ft.	3.50
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2058	Air Speed indicator 0 to 250 in 160 MPH	1.00
2059	Air Speed indicator 0 to 300 in 160 MPH	1.00
2060	Air Speed indicator 0 to 350 in 160 MPH	1.00
2061	Air Speed indicator 0 to 400 in 160 MPH	1.00
2062	Air Speed indicator 0 to 450 in 160 MPH	1.00
2063	Air Speed indicator 0 to 500 in 160 MPH	1.00
2064	Air Speed indicator 0 to 550 in 160 MPH	1.00
2065	Air Speed indicator 0 to 600 in 160 MPH	1.00
2066	Air Speed indicator 0 to 650 in 160 MPH	1.00
2067	Air Speed indicator 0 to 700 in 160 MPH	1.00
2068	Air Speed indicator 0 to 750 in 160 MPH	1.00
2069	Air Speed indicator 0 to 800 in 160 MPH	1.00
2070	Air Speed indicator 0 to 850 in 160 MPH	1.00
2071	Air Speed indicator 0 to 900 in 160 MPH	1.00
2072	Air Speed indicator 0 to 950 in 160 MPH	1.00
2073	Air Speed indicator 0 to 1000 in 160 MPH	1.00
2074	Air Speed indicator 0 to 1050 in 160 MPH	1.00
2075	Air Speed indicator 0 to 1100 in 160 MPH	1.00
2076	Air Speed indicator 0 to 1150 in 160 MPH	1.00
2077	Air Speed indicator 0 to 1200 in 160 MPH	1.00
2078	Air Speed indicator 0 to 1250 in 160 MPH	1.00
2079	Air Speed indicator 0 to 1300 in 160 MPH	1.00
2080	Air Speed indicator 0 to 1350 in 160 MPH	1.00
2081	Air Speed indicator 0 to 1400 in 160 MPH	1.00
2082	Air Speed indicator 0 to 1450 in 160 MPH	1.00
2083	Air Speed indicator 0 to 1500 in 160 MPH	1.00
2084	Air Speed indicator 0 to 1550 in 160 MPH	1.00
2085	Air Speed indicator 0 to 1600 in 160 MPH	1.00
2086	Air Speed indicator 0 to 1650 in 160 MPH	1.00
2087	Air Speed indicator 0 to 1700 in 160 MPH	1.00
2088	Air Speed indicator 0 to 1750 in 160 MPH	1.00
2089	Air Speed indicator 0 to 1800 in 160 MPH	1.00
2090	Air Speed indicator 0 to 1850 in 160 MPH	1.00
2091	Air Speed indicator 0 to 1900 in 160 MPH	1.00
2092	Air Speed indicator 0 to 1950 in 160 MPH	1.00
2093	Air Speed indicator 0 to 2000 in 160 MPH	1.00
2094	Air Speed indicator 0 to 2050 in 160 MPH	1.00
2095	Air Speed indicator 0 to 2100 in 160 MPH	1.00
2096	Air Speed indicator 0 to 2150 in 160 MPH	1.00
2097	Air Speed indicator 0 to 2200 in 160 MPH	1.00
2098	Air Speed indicator 0 to 2250 in 160 MPH	1.00
2099	Air Speed indicator 0 to 2300 in 160 MPH	1.00
2100	Air Speed indicator 0 to 2350 in 160 MPH	1.00
2101	Air Speed indicator 0 to 2400 in 160 MPH	1.00
2102	Air Speed indicator 0 to 2450 in 160 MPH	1.00
2103	Air Speed indicator 0 to 2500 in 160 MPH	1.00
2104	Air Speed indicator 0 to 2550 in 160 MPH	1.00
2105	Air Speed indicator 0 to 2600 in 160 MPH	1.00
2106	Air Speed indicator 0 to 2650 in 160 MPH	1.00
2107	Air Speed indicator 0 to 2700 in 160 MPH	1.00
2108	Air Speed indicator 0 to 2750 in 160 MPH	1.00
2109	Air Speed indicator 0 to 2800 in 160 MPH	1.00
2110	Air Speed indicator 0 to 2850 in 160 MPH	1.00
2111	Air Speed indicator 0 to 2900 in 160 MPH	1.00
2112	Air Speed indicator 0 to 2950 in 160 MPH	1.00
2113	Air Speed indicator 0 to 3000 in 160 MPH	1.00
2114	Air Speed indicator 0 to 3050 in 160 MPH	1.00
2115	Air Speed indicator 0 to 3100 in 160 MPH	1.00
2116	Air Speed indicator 0 to 3150 in 160 MPH	1.00
2117	Air Speed indicator 0 to 3200 in 160 MPH	1.00
2118	Air Speed indicator 0 to 3250 in 160 MPH	1.00
2119	Air Speed indicator 0 to 3300 in 160 MPH	1.00
2120	Air Speed indicator 0 to 3350 in 160 MPH	1.00
2121	Air Speed indicator 0 to 3400 in 160 MPH	1.00
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2123	Air Speed indicator 0 to 3500 in 160 MPH	1.00
2124	Air Speed indicator 0 to 3550 in 160 MPH	1.00
2125	Air Speed indicator 0 to 3600 in 160 MPH	1.00
2126	Air Speed indicator 0 to 3650 in 160 MPH	1.00
2127	Air Speed indicator 0 to 3700 in 160 MPH	1.00
2128	Air Speed indicator 0 to 3750 in 160 MPH	1.00
2129	Air Speed indicator 0 to 3800 in 160 MPH	1.00
2130	Air Speed indicator 0 to 3850 in 160 MPH	1.00
2131	Air Speed indicator 0 to 3900 in 160 MPH	1.00
2132	Air Speed indicator 0 to 3950 in 160 MPH	1.00
2133	Air Speed indicator 0 to 4000 in 160 MPH	1.00
2134	Air Speed indicator 0 to 4050 in 160 MPH	1.00
2135	Air Speed indicator 0 to 4100 in 160 MPH	1.00
2136	Air Speed indicator 0 to 4150 in 160 MPH	1.00
2137	Air Speed indicator 0 to 4200 in 160 MPH	1.00
2138	Air Speed indicator 0 to 4250 in 160 MPH	1.00
2139	Air Speed indicator 0 to 4300 in 160 MPH	1.00
2140	Air Speed indicator 0 to 4350 in 160 MPH	1.00
2141	Air Speed indicator 0 to 4400 in 160 MPH	1.00
2142	Air Speed indicator 0 to 4450 in 160 MPH	1.00
2143	Air Speed indicator 0 to 4500 in 160 MPH	1.00
2144	Air Speed indicator 0 to 4550 in 160 MPH	1.00
2145	Air Speed indicator 0 to 4600 in 160 MPH	1.00
2146	Air Speed indicator 0 to 4650 in 160 MPH	1.00
2147	Air Speed indicator 0 to 4700 in 160 MPH	1.00
2148	Air Speed indicator 0 to 4750 in 160 MPH	1.00
2149	Air Speed indicator 0 to 4800 in 160 MPH	1.00
2150	Air Speed indicator 0 to 4850 in 160 MPH	1.00
2151	Air Speed indicator 0 to 4900 in 160 MPH	1.00
2152	Air Speed indicator 0 to 4950 in 160 MPH	1.00
2153	Air Speed indicator 0 to 5000 in 160 MPH	1.00
2154	Air Speed indicator 0 to 5050 in 160 MPH	1.00
2155	Air Speed indicator 0 to 5100 in 160 MPH	1.00
2156	Air Speed indicator 0 to 5150 in 160 MPH	1.00
2157	Air Speed indicator 0 to 5200 in 160 MPH	1.00
2158	Air Speed indicator 0 to 5250 in 160 MPH	1.00
2159	Air Speed indicator 0 to 5300 in 160 MPH	1.00
2160	Air Speed indicator 0 to 5350 in 160 MPH	1.00
2161	Air Speed indicator 0 to 5400 in 160 MPH	1.00
2162	Air Speed indicator 0 to 5450 in 160 MPH	1.00
2163	Air Speed indicator 0 to 5500 in 160 MPH	1.00
2164	Air Speed indicator 0 to 5550 in 160 MPH	1.00
2165	Air Speed indicator 0 to 5600 in 160 MPH	1.00
2166	Air Speed indicator 0 to 5650 in 160 MPH	1.00
2167	Air Speed indicator 0 to 5700 in 160 MPH	1.00
2168	Air Speed indicator 0 to 5750 in 160 MPH	1.00
2169	Air Speed indicator 0 to 5800 in 160 MPH	1.00
2170	Air Speed indicator 0 to 5850 in 160 MPH	1.00
2171	Air Speed indicator 0 to 5900 in 160 MPH	1.00
2172	Air Speed indicator 0 to 5950 in 160 MPH	1.00
2173	Air Speed indicator 0 to 6000 in 160 MPH	1.00
2174	Air Speed indicator 0 to 6050 in 160 MPH	1.00
2175	Air Speed indicator 0 to 6100 in 160 MPH	1.00
2176	Air Speed indicator 0 to 6150 in 160 MPH	1.00
2177	Air Speed indicator 0 to 6200 in 160 MPH	1.00
2178	Air Speed indicator 0 to 6250 in 160 MPH	1.00
2179	Air Speed indicator 0 to 6300 in 160 MPH	1.00
2180	Air Speed indicator 0 to 6350 in 160 MPH	1.00
2181	Air Speed indicator 0 to 6400 in 160 MPH	1.00
2182	Air Speed indicator 0 to 6450 in 160 MPH	1.00
2183	Air Speed indicator 0 to 6500 in 160 MPH	1.00
2184	Air Speed indicator 0 to 6550 in 160 MPH	1.00
2185	Air Speed indicator 0 to 6600 in 160 MPH	1.00
2186	Air Speed indicator 0 to 6650 in 160 MPH	1.00
2187	Air Speed indicator 0 to 6700 in 160 MPH	1.00
2188	Air Speed indicator 0 to 6750 in 160 MPH	1.00
2189	Air Speed indicator 0 to 6800 in 160 MPH	1.00
2190	Air Speed indicator 0 to 6850 in 160 MPH	1.00
2191	Air Speed indicator 0 to 6900 in 160 MPH	1.00
2192	Air Speed indicator 0 to 6950 in 160 MPH	1.00
2193	Air Speed indicator 0 to 7000 in 160 MPH	1.00
2194	Air Speed indicator 0 to 7050 in 160 MPH	1.00
2195	Air Speed indicator 0 to 7100 in 160 MPH	1.00
2196	Air Speed indicator 0 to 7150 in 160 MPH	1.00
2197	Air Speed indicator 0 to 7200 in 160 MPH	1.00
2198	Air Speed indicator 0 to 7250 in 160 MPH	1.00
2199	Air Speed indicator 0 to 7300 in 160 MPH	1.00
2200	Air Speed indicator 0 to 7350 in 160 MPH	1.00
2201	Air Speed indicator 0 to 7400 in 160 MPH	1.00
2202	Air Speed indicator 0 to 7450 in 160 MPH	1.00
2203	Air Speed indicator 0 to 7500 in 160 MPH	1.00
2204	Air Speed indicator 0 to 7550 in 160 MPH	1.00
2205	Air Speed indicator 0 to 7600 in 160 MPH	1.00
2206	Air Speed indicator 0 to 7650 in 160 MPH	1.00
2207	Air Speed indicator 0 to 7700 in 160 MPH	1.00
2208	Air Speed indicator 0 to 7750 in 160 MPH	1.00
2209	Air Speed indicator 0 to 7800 in 160 MPH	1.00
2210	Air Speed indicator 0 to 7850 in 160 MPH	1.00
2211	Air Speed indicator 0 to 7900 in 160 MPH	1.00
2212	Air Speed indicator 0 to 7950 in 160 MPH	1.00
2213	Air Speed indicator 0 to 8000 in 160 MPH	1.00
2214	Air Speed indicator 0 to 8050 in 160 MPH	1.00
2215	Air Speed indicator 0 to 8100 in 160 MPH	1.00
2216	Air Speed indicator 0 to 8150 in 160 MPH	1.00
2217	Air Speed indicator 0 to 8200 in 160 MPH	1.00
2218	Air Speed indicator 0 to 8250 in 160 MPH	1.00
2219	Air Speed indicator 0 to 8300 in 160 MPH	1.00
2220	Air Speed indicator 0 to 8350 in 160 MPH	1.00
2221	Air Speed indicator 0 to 8400 in 160 MPH	1.00
2222	Air Speed indicator 0 to 8450 in 160 MPH	1.00
2223	Air Speed indicator 0 to 8500 in 160 MPH	1.00
2224	Air Speed indicator 0 to 8550 in 160 MPH	1.00
2225	Air Speed indicator 0 to 8600 in 160 MPH	1.00
2226	Air Speed indicator 0 to 8650 in 160 MPH	1.00
2227	Air Speed indicator 0 to 8700 in 160 MPH	1.00
2228	Air Speed indicator 0 to 8750 in 160 MPH	1.00
2229	Air Speed indicator 0 to 8800 in 160 MPH	1.00
2230	Air Speed indicator 0 to 8850 in 160 MPH	1.00
2231	Air Speed indicator 0 to 8900 in 160 MPH	1.00
2232	Air Speed indicator 0 to 8950 in 160 MPH	1.00
2233	Air Speed indicator 0 to 9000 in 160 MPH	1.00
2234	Air Speed indicator 0 to 9050 in 160 MPH	1.00
2235	Air Speed indicator 0 to 9100 in 160 MPH	1.00
2236	Air Speed indicator 0 to 9150 in 160 MPH	1.00
2237	Air Speed indicator 0 to 9200 in 160 MPH	1.00
2238	Air Speed indicator 0 to 9250 in 160 MPH	1.00
2239	Air Speed indicator 0 to 9300 in 160 MPH	1.00
2240	Air Speed indicator 0 to 9350 in 160 MPH	1.00
2241	Air Speed indicator 0 to 9400 in 160 MPH	1.00
2242	Air Speed indicator 0 to 9450 in 160 MPH	1.00
2243	Air Speed indicator 0 to 9500 in 160 MPH	1.00
2244	Air Speed indicator 0 to 9550 in 160 MPH	1.00
2245	Air Speed indicator 0 to 9600 in 160 MPH	1.00
2246	Air Speed indicator 0 to 9650 in 160 MPH	1.00
2247	Air Speed indicator 0 to 9700 in 160 MPH	1.00
2248	Air Speed indicator 0 to 9750 in 160 MPH	1.00
2249	Air Speed indicator 0 to 9800 in 160 MPH	1.00
2250	Air Speed indicator 0 to 9850 in 160 MPH	1.00
2251	Air Speed indicator 0 to 9900 in 160 MPH	1.00
2252	Air Speed indicator 0 to 9950 in 160 MPH	1.00
2253	Air Speed indicator 0 to 10000 in 160 MPH	1.00
2254	Air Speed indicator 0 to 10050 in 160 MPH	1.00
2255	Air Speed indicator 0 to 10100 in 160 MPH	1.00
2256	Air Speed indicator 0 to 10150 in 160 MPH	1.00
2257	Air Speed indicator 0 to 10200 in 160 MPH	1.00
2258	Air Speed indicator 0 to 10250 in 160 MPH	1.00
2259	Air Speed indicator 0 to 10300 in 160 MPH	1.00
2260	Air Speed indicator 0 to 10350 in 160 MPH	1.00
2261	Air Speed indicator 0 to 10400 in 160 MPH	1.00
2262	Air Speed indicator 0 to 10450 in 160 MPH	1.00
2263	Air Speed indicator 0 to 10500 in 160 MPH	1.00
2264	Air Speed indicator 0 to 10550 in 160 MPH	1.00
2265	Air Speed indicator 0 to 10600 in 160 MPH	1.00
2266	Air Speed indicator 0 to 10650 in 160 MPH	1.00
2267	Air Speed indicator 0 to 10700 in 160 MPH	1.00
2268	Air Speed indicator 0 to 10750 in 160 MPH	1.00
2269	Air Speed indicator 0 to 10800 in 160 MPH	1.00
2270	Air Speed indicator 0 to 10850 in 160 MPH	1.00
2271	Air Speed indicator 0 to 10900 in 160 MPH	1.00
2272	Air Speed indicator 0 to 10950 in 160 MPH	1.00
2273	Air Speed indicator 0 to 11000 in 160 MPH	1.00
2274	Air Speed indicator 0 to 11050 in 160 MPH	1.00
2275	Air Speed indicator 0 to 11100 in 160 MPH	1.00
2276	Air Speed indicator 0 to 11150 in 160 MPH	1.00
2277	Air Speed indicator 0 to 11200 in 160 MPH	1.00
2278	Air Speed indicator 0 to 11250 in 160 MPH	1.00
2279	Air Speed indicator 0 to 11300 in 160 MPH	1.00
2280	Air Speed indicator 0 to 11350 in 160 MPH	1.00
2281	Air Speed indicator 0 to 11400 in 160 MPH	1.00
2282	Air Speed indicator 0 to 11450 in 160 MPH	1.00
2283	Air Speed indicator 0 to 11500 in 160 MPH	1.00
2284	Air Speed indicator 0 to 11550 in 160 MPH	1.00
2285	Air Speed indicator 0 to 11600 in 160 MPH	1.00
2286	Air Speed indicator 0 to 11650 in 160 MPH	1.00
2287	Air Speed indicator 0 to 11700 in 160 MPH	1.00
2288	Air Speed indicator 0 to 11750 in 160 MPH	1.00
2289	Air Speed indicator 0 to 11800 in 160 MPH	1.00
2290	Air Speed indicator 0 to 11850 in 160 MPH	1.00
2291	Air Speed indicator 0 to 11900 in 160 MPH	1.00
2292	Air Speed indicator 0 to 11950 in 160 MPH	1.00
2293	Air Speed indicator 0 to 12000 in 160 MPH	1.00
2294	Air Speed indicator 0 to 12050 in 160 MPH	1.00
2295	Air Speed indicator 0 to 12100 in 160 MPH	1.00
2296	Air Speed indicator 0 to 12150 in 160 MPH	1.00
2297	Air Speed indicator 0 to 12200 in 160 MPH	1.00
2298	Air Speed indicator 0 to 12250 in 160 MPH	1.00
2299	Air Speed indicator 0 to 12300 in 160 MPH	1.00
2300	Air Speed indicator 0 to 12350 in 160 MPH	1.00
2301	Air Speed indicator 0 to 12400 in 160 MPH	1.00
2302	Air Speed indicator 0 to 12450 in 160 MPH	1.00
2303	Air Speed indicator 0 to 12500 in 160 MPH	1.00
2304	Air Speed indicator 0 to 12550 in 160 MPH	1.00
2305	Air Speed indicator 0 to 12600 in 160 MPH	1.00
2306	Air Speed indicator 0 to 12650 in 160 MPH	1.00
2307	Air Speed indicator 0 to 12700 in 160 MPH	1.00
2308	Air Speed indicator 0 to 12750 in 160 MPH	1.00
2309	Air Speed indicator 0 to 12800 in 160 MPH	1.00
2310	Air Speed indicator 0 to 12850 in 160 MPH	1.00
2311	Air Speed indicator 0 to 12900 in 160 MPH	1.00
2312	Air Speed indicator 0 to 12950 in 160 MPH	1.00
2313	Air Speed indicator 0 to 13000 in 160 MPH	1.00
2314	Air Speed indicator 0 to 13050 in 160 MPH	1.00
2315	Air Speed indicator 0 to 13100 in 160 MPH	1.00
2316	Air Speed indicator 0 to 13150 in 160 MPH	1.00
2317	Air Speed indicator 0 to 13200 in 160 MPH	1.00
2318	Air Speed indicator 0 to 13250 in 160 MPH	1.00
2319	Air Speed indicator 0 to 13300 in 160 MPH	1.00
2320	Air Speed indicator 0 to 13350 in 160 MPH	1.00
2321	Air Speed indicator 0 to 13400 in 160 MPH	1.00
2322	Air Speed indicator 0 to 13450 in 160 MPH	1.00
2323	Air Speed indicator 0 to 13500 in 160 MPH	1.00
2324	Air Speed indicator 0 to 13550 in 160 MPH	1.00
2325	Air Speed indicator 0 to 13600 in 160 MPH	1.00
2326	Air Speed indicator 0 to 13650 in 160 MPH	1.00
2327	Air Speed indicator 0 to 13700 in 160 MPH	1.00
2328	Air Speed indicator 0 to 13750 in 160 MPH	1.00
2329	Air Speed indicator 0 to 13800 in 160 MPH	1.00
2330	Air Speed indicator 0 to 13850 in 160 MPH	1.00
2331	Air Speed indicator 0 to 13900 in 160 MPH	1.0

*What are you going to do this summer ?*



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